

Enabling Collision Avoidance with Raw Position Measurements and Updated ADS-B Software

Raw GPS measurements including carrier phase and pseudorange are used to derive the aircraft position broadcast via ADS-B. But broadcasting the raw data instead enables orders-of-magnitude improvement in flight path projections, reducing the risk of collisions in skies soon to be crowded with UAVs. This approach could also improve operations in GPS-impaired environments and the integration of non-GPS data sources. The technique is so promising the global standards group SAE International has begun developing standards to support it. That work is underway in the newly launched SMCPNT technical committee, which will reach across sectors as part of SAE's new Systems Management Council.

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ollision avoidance will be more practically and universally achievable, even in skies crowded with unmanned aerial vehicles (UAVs), if the aviation community takes advantage of the raw measurements already present in today's GPS receivers, but largely ignored in favor of using GPS position coordinates. Changes to existing aviation equipment could enable aircraft to estimate the flight paths of other aircraft far more accurately enabling safer operations even under impaired conditions. The use of the raw data, once integrated into standardized flight protocols, could dramatically help prevent midair accidents even if one or both of the aircraft is unmanned.

This flight-validated approach uses established algorithms, readily available universal access transceivers (UATs) and existing communication message formats (as described by P. Duan et alia in Additional Resources). There is one essential departure from current practice: including the raw pseudorange and carrier phase data within the automatic dependent surveillancebroadcast (ADS-B) messages in place of the derived position coordinates. This same approach could be applied with great benef t should a separate system, similar to ADS-B, need to be developed to support UAV operations.

For more than 50 years it has been feasible to combine intermittent partial data - of dif erent types at varying accuracies with different sensitivities from different directions at different times - and extract all benef t therein. The seemingly unspectacular step of using raw measurements in the message opens the door to using powerful, widely understood methods for predicting flight paths — and therefore the points of potential collision — and handling situations where position determination is hampered due to jamming or because there are not enough satellites in view. It should be possible to do all of this for aircraft at various altitudes without building and certifying extensive new radar or other ground infrastructure.

This paper primarily involves GPS and airborne operation, but a halfcentury of experience combining data enables this technique to be dramatically extended. Integration of dif erent sensors (eLoran, DME, etc.) is straightforward; a claim that has been verif ed and documented.

Raw Measurements Improve Estimates

For a host of reasons, techniques using raw measurements — which are present in any navigation sensor — will outperform by orders of magnitude techniques relying on position reporting. Dif erential GPS (DGPS) owes its spectacular success to its use of raw measurements. A Kalman tracker uses weights based on an extensive array of data. There are across-axis correlations between error components in different directions, between components of position and velocity, etc. — and the sensitivity of each individual observation to every one of those components is taken into account.

Unfortunately *none* of those features can be used when starting with coordinates *derived from* raw measurements. Since ADS-B link bandwidth can't hold its existing content *plus* all that correlation information, ADS-B messages contain *no* correlations but airborne computation armed with a history of raw measurements can deduce *all*. The contrast could hardly be more compelling.

The extended squitter message can remain unchanged except for the replacement of position and velocity by raw measurements. Since navigation systems commonly allow multiple message types, however, position reports are not strictly ruled out. Occasionally another message type could be used to broadcast the position for, say, track f le initiation. To realize the performance potential, however, messages containing raw measurement data would be far more frequent.

But why go to the trouble of changing an established methodology (that is position reporting) for one based on Kalman filtering considerations? There are multiple reasons. Probably most obvious, instantaneous position is fleeting for anything airborne; data must be combined. Even satellite navigation uses several observations to get a position f x (i.e., four satellites in air). Another reason stems from the nature of the position that must be determined: To support collision avoidance the position information for one object must be determined relative to the other objects in nearby space and projected into the future. Using raw data makes these calculations far, far more accurate.

Consider, for example, a pair of position coordinates, one with perfect longitude but a kilometer of error in the north direction, and the other with exact latitude but its east/west position is of by a kilometer. Averaging them gives "only" 500 meters of error in both!

Even if different tolerances of different position reports are taken into account, ignoring wide variations in sensitivity and correlation parameters is ruinous. There are no tight velocity accuracy requirements specif ed for ADS-B as a result.

Accurate Velocity Essential

In fact the velocity requirement for ADS-B is loose in multiple ways. For characteristics that matter in regard to collision avoidance, velocity is a vector - a vector *relative* to other objects in nearby space regardless of those objects' latitude or longitude - and with errors having statistical properties. Error values of several meters/second, even as much as 10 meters/second, have been published in connection with ADS-B. A more subtle point is that even a substantially lower 1 meter/second error value is dangerous statistically. Without detailed elaboration, this much needs to be recognized: extreme value theory (EVT) shows that, even if all errors were Gaussian, *mixed* Gaussian probability of ers far less assurance than intuition would suggest. Instances of exceeding 10 sigma cannot be discounted (see Farrell J., and F. van Graas, Additional Resources). To ignore that is to accept an excessive and unsafe risk; "unlikely" is often not unlikely enough. A 10 meter/second error is therefore not too farfetched to consider. A sequence of position reports can suffice for transoceanic flight but not within crowded airspace.

Avoiding mid-air crashes requires a fresh look at system priorities. For example it is *not* critical to have highly accurate position reports. For airliners with wingspans tens of meters long moving at hundreds of kilometers per hour, precise position is fleeting and unnecessary. A few meters of current position error will be insignif cant and working to ref ne that position error would be pointless.

It is essential, however, to have highly accurate velocities. The product of (velocity error) × (time to closest approach) is dominant when it comes to

COLLISION AVOIDANCE

The Advantages of Raw Measurements

A 2012 flight validation used GPS without augmenting system corrections but with raw measurements from receivers. Twenty years earlier Lincoln Labs successfully demonstrated GPS broadcasts with Mode S beacon transponders at Logan Airport as described in E. T. Bayliss et alia. (Note: the 2012 flight in the f rst case used UATs instead of Mode S). Transmitted positions enabled each participant to track every other participant's data while minimizing or eliminating garble, by replacing conventional interrogations with information in assigned time slots (as ADS-B currently prescribes).

One basic modif cation of the Lincoln Labs methodology was advocated in the work by J. Farrell and E. McConkey and linked with another system — the Joint Tactical Information Distribution System (JTIDS) to form a more general application. Instead of coordinates, the transmitted message's 48 information bits can contain *raw uncorrected* measurements. Data compression and the cycling of in-view GPS satellites can mitigate bandwidth limitations.

The introductory paragraphs of an earlier article titled "Send measurements, not coordinates," (J. Farrell et alia) noted eight crucial advantages. In combination with each tracking-every-other feature already noted — a later expansion of that paper (J. Farrell and M. Farrell, Additional Resources) of ered an even more extensive list of advantages:

•

f x

•

Ability to resolve conflicts

by turns or speed change

Applicability to both 3-D

(in-air) and 2-D (runway

Removal of potential dan-

ger in the event of datum

reference nonuniformity

Full usage of available

data when too few satel-

lites are visible for a full

Integrity checks enabled

with any number of satel-

Unrestricted algo-

rithm release (no strings

attached or proprietary

No need for augmentation

(corrections) from ground

Opportunity for partici-

pants to share observa-

tions of nonparticipants

Retention of applicability

with or without prospec-

Insensitivity to different

models used in different

tive modernizations

constellations

lites observed

claims)

stations

instead of climb/dive

incursion) encounters

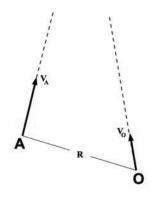
- Two decades of stunningly successful differential GPS operations demonstrate this approach
- Error source cancellation capability is intrinsic to dif erential GPS
- The ability to account for specific sensitivities of each individual measurement
- The opportunity to employ those sensitivities to assign data weighting adaptively
- Widely known techniques for minimization of statistical error resulting from that adaptivity
- Prompt determination of full information (cross-range as well as along range)
- Presence in that information of accurate dynamics as well as current position
- Ability to use the dynamics to anticipate time of closest approach
- Ability to deduce, from the dynamics, the miss distance at that future time
- These benef ts are utterly absent if calculations must rely only on instantaneous position reports.

collision avoidance. Instead of meters per second velocity accuracy, airliners need centimeters/second accuracy; otherwise the projected position over time is so crude as to be useless for collision avoidance.

Collision avoidance demands assurance of suf cient distance at time of closest approach. That clearly requires accurate knowledge of velocity — specif cally the relative velocity vector. Stitching coordinates together from reports of latitude + longitude + altitude ("LLH") cannot deliver that, which explains why ADS-B does not promise good velocity. Errors of 10 meters/second have been published with little elaboration (not relative, not vectorial, and with no statistical boundaries). If the closest approach is a minute away, then even the most elementary arithmetic assigns 600 meters of uncertainty to that future position.

Consider, for example, two aircraft flying at the same altitude: "Ownship" at location **O** with velocity **VO** and "Anothership" at location **A** with velocity **VA**. They are instantaneously separated by vector **R** which, for closing scenarios, is shrinking. The closest approach will occur at time *T* when the component of *relative* velocity (vector dif erence **VA** - **VO**, not shown) parallel to **R** passes through zero — the perpendicular component is miss distance. That simple scenario has appeared in countless context-dependent forms (e.g., with intruder at **A** and evader at **O** in or with target at **A** and an interceptor or projectile at **O** in military operations). Determination of *T* and minimum separation distance follows easily from relations just stated, readily superseded whenever maneuvers subsequently change either velocity.

Now to conf rm the case for precise velocity: rather than *current* position, collision avoidance requires accurate *future*



position (i.e., at time *T*). When position is projected *T* seconds ahead, a *10* meter/second velocity error will cause that predicted future position to be in error by *10 T* meters. With that much error in each of two horizontal axes that product will be squared, producing an unacceptably large area of uncertainty. Trying to steer away from an unknown place has no meaning. The Traffic

Collision Avoidance System (TCAS) uses climb/dive maneuvers instead. Imagine that becoming a commonplace event as the skies f ll with unmanned aircraft.

There are many facets to this subject, but the simple, f xedaltitude case above is useful for establishing some fundamentals:

- Time *T* used above matches the "tau" of the traf *c* collision avoidance system (TCAS) *only* on a collision course
- Ef ects of current position error matter far less than velocity error

- Centimeter/second accuracy for velocity rather than meters/ second can enable horizontal evasive strategies
- Longer values of *T* (earlier evasive action) is also thereby made possible
- Earlier evasive action is highly preferable to TCAS's abrupt violent maneuvering
- TCAS cannot act early because valid decisions require accurate tracks
- TCAS tracks are informed by accurate range but very *crude* crossrange data
- TCAS crossrange information improves only as the sightline rotates
- Sightline rotation increases at close range exactly the *waterloo* for collision avoidance!
- The precise satellite navigation data used in the previously mentioned P. Daun *et alia* article (Additional Resources) provides full 3-D tracks quickly
- Requisite speed changes and *T* have been quantif ed for many cases (see Farrell, J., "Collision avoidance by speed change," Additional Resources).

Raw Measurements Improve Tracking

As noted in the article on airport surface surveillance (J. Farrell and E. McConkey, Additional Resources) computers now can

easily maintain integrated track f les for every participant in any scenario. Even in the 1970s two missiles plus two aircraft were simultaneously tracked in real time with an electronically steered radar antenna at White Sands. The estimation algorithms in the White Sands case were fed by raw observations (range, azimuth and elevation in that case) — *never* with coordinate pseudo-measurements — and tracking from high dynamic platforms with "Ownship" navigation is a straightforward extension of tracking from a stationary location.

Today's computing capabilities readily enable each participant to maintain a bank of extended Kalman f lters (EKFs) with a separate track f le for each participant and with every participant having a designated slot in the sequence of transmitted messages from all the participants. The full set of participants should include every object that could be involved in any collision. The track f le in any participant's database is not tied to coordinates; it's scalar. From those scalars each participant can construct a set of vectors and all those vectors will be correct and can be expressed in his own perceived reference. If that perception dif ers from the other participants' (due to misalignments or even a dif erent datum), performance does not suf er one iota.

Air-to-air tracking has always placed the Ownship described in the example above at the center of its "own little

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Integrity Testing: Ultra-simple and Rigorously Validated

Volumes have been written on Receiver Autonomous Integrity Monitoring (RAIM), often supported by sophisticated analytical methods and substantial mathematical development. The good news is the hard work has been done. All a program needs is a set of expressions to put into code. Even more fortuitous is further simplif cation of those expressions—and that also has been done. Moreover, the way that simplif cation has been done allows extension beyond GNSS, to include every morsel of data used for navigation.

Conventional RAIM uses f ve satellites for fault detection and six satellites for fault exclusion or isolation. Because every subset of four within those sets must support adequate geometric dilution of precision (GDOP), exclusion or isolation is not always available. Then, when the f ve-satellite detection indicates excessive error, conventional RAIM rejects the whole quintet, the good along with the bad. Forcing valid data to suf er from "guilt-by-association" is extremely wasteful and unnecessary. <http://jameslfarrell.com/single-measurementraim/> Reversing the loss is especially urgent when data availability is marginal. A variety of advanced integrity features of ers:

- Addition of cyclic bias estimation without changing navigation solutions
- Circumvention of parity vector operations added for conventional fault isolation/exclusion
- Replacement of that parity vector by a parity scalar with no loss of capability
- Normalization of that parity scalar to a form with variance equal to one (dimensionless)
- Accounting for ef ects of correlations incurred by dif erencing
- Inclusion of closed form matrix solutions for fault detection and isolation/ exclusion with correlations

- Extension to separate validation of each individual measurement, whether others are present or not
- Opportunity to verify single-measurement tests when multi-satellite isolation/exclusion is feasible
- Support by rigorous theory (matrix decomposition etc.) with no need to employ it in operation.
- The normalized parity scalar test for every individual measurement (everyone understands a dimensionless scalar random variable with sigma = 1) provides a vital means of operating with any and every available source of navigation information.

world" without any degradation. What matters is *relative* state (position, velocity...) in Ownship's "own little world" expressed and maintained consistently the same way for all. Sharing satellite navigation data with others will not introduce any error since those measurements are scalar —unattached to any coordinate frame. If the presence of one participant with overriding authority must be identif ed, one of the participants *could* be a tower. With the exception of the tower, if there is one, moving participants would make path adjustments with each message received as the scenario unfolds. Those smaller, repeated adjustments over time will prove far less abrupt than making a start-from-scratch change at close range.

Dealing With Too Few Satellites

Using the raw data also enables the development of track f les in situations where there are not enough GPS satellites in view. In fact, in some urban canyon scenarios it is possible to have situations where there are never enough satellites available with good enough geometries.

As things now stand, if an aircraft's GNSS receiver does not have enough satellites in view it is not able to determine its position and therefore has nothing to broadcast on ADS-B. That is a scandalous waste of very accurate information. Raw data measured every second or so will give you a far better track f le than the usage of GPS coordinates. Stitching coordinates together to get velocity gives totally inadequate performance. That is why ADS-B, even with all the ADS-B Out and ADS-B In information, will not provide accurate velocity.

UAV-Specific Considerations

While UAVs will be responsible for taking evasive action, they will be less burdened in other respects. Their lower speed af ords multiple advantages: more time for evasion, track f le initiation at short range (allowing operation at low power) and the ability to make tighter turns. All of these factors make sense and avoid easier for UAVs than it is for fast-moving airliners.

Nor will UAVs require the sophistication used by P. Duan *et alia* in Additional Resources, which used 1-second changes in meticulously prepared carrier phase measurements. Many satellite navigation receivers don't use carrier phase but all have pseudoranges — those will suf ce as long as they are made available with appropriate time stamps. Also, decimeters/second rather than centimeter/second velocity error will be acceptable — again because of a UAV's slower speed. With evasion by acceleration or deceleration, for example, the simple program (see J. Farrell, "Collision avoidance by speed change," in Additional Resources) can just have dif erent parameters. Finally, evasion strategy won't be limited to speed changes; descent or turns can be used in some circumstances.

The Challenge

Though the advantages of using raw measurement are clear, change is not easy. Using position, and over recent decades GPS-derived position, in ADS-B messaging has long been the established approach. However the integration of unmanned aircraft is such a monumental challenge that new techniques and air traf c management systems for UAVs are being considered. Incorporating raw measurements not only of ers a capability that supports safe UAV integration, but of ers real advantages to manned flight operations as well — and there is a rock-solid track record supporting both double dif erencing (see "Double Dif erencing" sidebar, page 52) and all modes of tracking (air-to-air, air-to-surface, surface-to-air, surface-to-surface) by adaptive modern estimation.

Conclusions and Recommendations

Working with the raw measurements instead of relying only on the position calculated using those measurements makes it possible to apply the techniques that made dif erential GPS so spectacularly successful. This approach also opens the door for the integration of data from information sources completely dif erent from GNSS and from each other. Raw measurements of er the only way to achieve true integration with systems like DME, eLoran and Iridium and, especially for cooperating UAVs, signals-of-opportunity (see R. Kapoor et *alia*, Additional Resources).The scope can also be extended to include observations of nonparticipants (see Fig. 9.4 in J. Farrell, "GNSS Aided Navigation and Tracking – Inertially Augmented or Autonomous," in Additional Resources).

The improvements in situational awareness are dramatic enough to suggest redefining availability and continuity of operation. Less obvious but equally decisive is how this approach strengthens integrity. Every individual measurement can be acceptance-tested — directly, easily, and independently of all others, supported by demonstrated equivalence to rigorous, widely accepted parity methods (see "Integrity Testing: Ultra-simple And Rigorously Validated" sidebar, page 50).

These dramatic improvements do not require new discoveries or the invention of new equipment. A revision of the ADS-B message content, hopefully via a software update and the inclusion of raw measurements in any new system developed to support UAVs — will enable a host of spectacular benef ts from already readily available data. In fact the use of raw measurements is so promising that SAE International has begun developing standards to bring this approach into the mainstream.

There also are documented *non-proprietary* navigation algorithms already available that make it possible to tap the value of the raw measurements. These algorithms could help keep costs down and speed the launch of a pilot project to test this approach, especially in the case of unmanned aircraft. There is enormous commercial, political and regulatory pressure to integrate UAVs into the national airspace. A pilot project could support both manned and unmanned aviation by strengthening reliability and robustness while boosting accuracy and integrity — thereby helping keep aircraft out of each other's way.

An old movie scene showed Bob Hope trudging through a desert, desperately uttering "water, water" — then f nding himself waist deep in a stream moments later, mumbling "mirage, mirage." The advantages of using raw measurements for ADS-B and systems similar to ADS-B are not a mirage. Between what we know and what we do is a wide gulf. Let's close it.

Appendix—Additional Topics

Two separate but related articles from a recent Institute of Navigation newsletter <http://www.ion.org/publications/upload/ v26n3.pdf > discuss important developments in GPS/GNSS interfacing. Starting on page 1 and continued on page 7, the



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Double Differencing

Ignoring wide variations in sensitivity and correlation parameters is ruinous, but it is possible to recover that information.

In August 2000 I presented the rawmeasurements-in-squitter-messages concept as a natural extension of GPS double dif erencing and asked RTCA SC186WG4 members to imagine two happenings:

- Let every system and every plan in existence be only supplemental/ backup
- Let every participant compare his own data from each separate satellite to corresponding measurements from all other participants, weighting every individual dif erence adaptively according to its information content (we've been optimizing partial information weights for a half century).

A rock solid track record supports double dif erencing and all modes of tracking (air-to-air, air-to- surface, surface-to-air, surface-to-surface) by modern estimation. A sequence of position reports can suf ce for transoceanic flight but **not** within crowded airspace. As noted in [4] computerized "bookkeeping" can easily maintain track f les for every participant in any scenario.

first describes major improvements in Android handsets. The second, on pages 14-15, announces formation of a Society of Automotive Engineers (SAE) International working group, which will work on the standards cited in the Conclusions and Recommendations section of this article, ensuring the extension of benef ts to the vast majority of devices. (SAE International is a global association of more than 128,000 engineers and related technical experts in the aerospace, automotive and commercial-vehicle industries.) These were preceded by other publications emphasizing the benefits offered by working with measurement data. One, more than 25 years old (J. Farrell and F. van Graas, Additional Resources) was in fact preceded by an obscure (1977) NAE-CON paper. Two more recent videos <https://youtube/10RCAY-B9mk> and <https://www.youtube.com/watch?v=2 X88s4o74c4&list=UUSphzH7ReVjg0-Wh3pw0ZFA&index=10> plus a presentation <http://www.gps.gov/governance/ advisory/meetings/2015-06/farrell.pdf> of er additional background.

The centimeter/second residuals achieved in flight test described previously by P. Daun et alia, in Additional Resources, were obtained by using sequential changes in carrier phase measurements measured once a second. Unlike the carrier phases themselves, 1-second changes in them are interoperable (i.e., regardless of dif erent timing and/or geoid conventions used for separate constellations) and immune to catastrophic error (see links to https:// jameslfarrell.com content in Additional Resources). Furthermore, because two main sources of propagation error change very little over a second, there is no need for a mask angle — a trait that benef ts geometric dilution of precision (GDOP) for velocity.

Additional Resources

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